



JUNO ORBIT DETERMINATION EXPERIENCE DURING FIRST YEAR AT JUPITER

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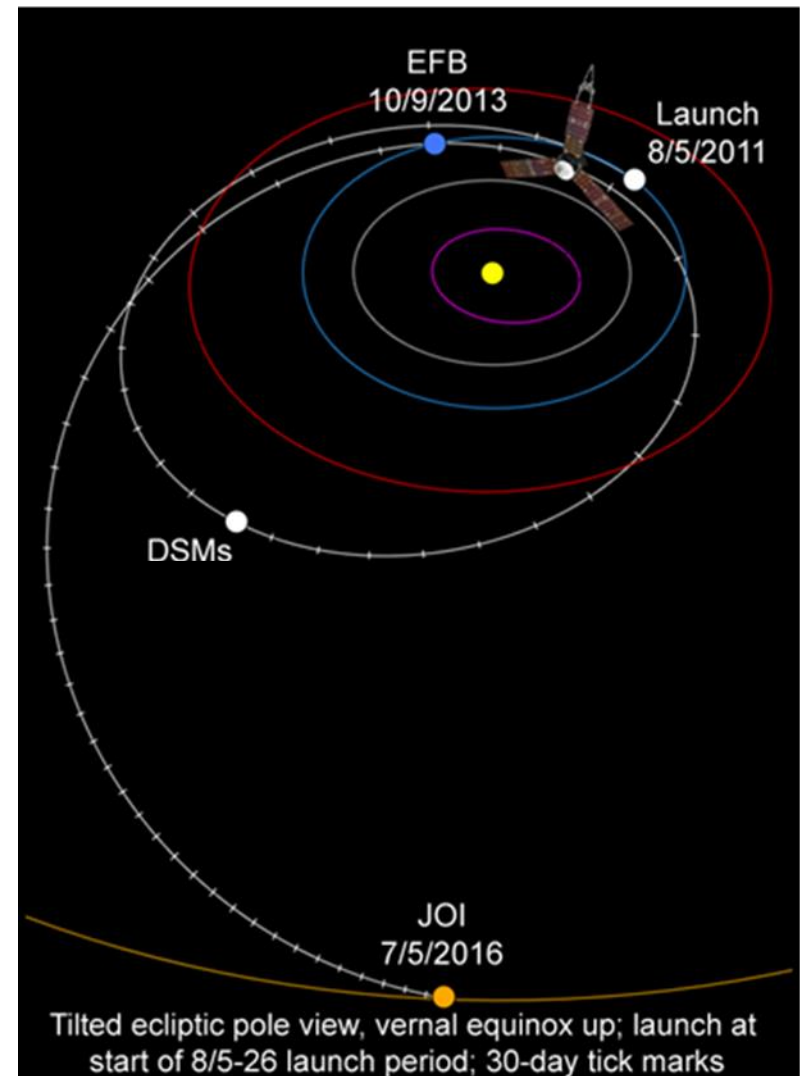
Introduction

- **NASA New Frontiers Mission**
- **Launched: 05-Aug-2011** **JOI: 05-Jul-2016**
 - Included 500-km altitude, Earth gravity assist (EGA) to gain energy needed to reach Jupiter
 - Science phase at Jupiter of highly elliptical, near-polar 53-day orbits
- **Payload of nine instruments:**
 - *Determine how much water is in Jupiter's atmosphere, which helps determine which planet formation theory is correct (or if new theories are needed)*
 - *Look deep into Jupiter's atmosphere to measure composition, temperature, cloud motions and other properties*
 - *Map Jupiter's magnetic and gravity fields, revealing the planet's deep structure*
 - *Explore and study Jupiter's magnetosphere near the planet's poles, especially the auroras – Jupiter's northern and southern lights – providing new insights about how the planet's enormous magnetic force field affects its atmosphere.*

Maneuvers: Launch to Earth Gravity Assist (EGA)



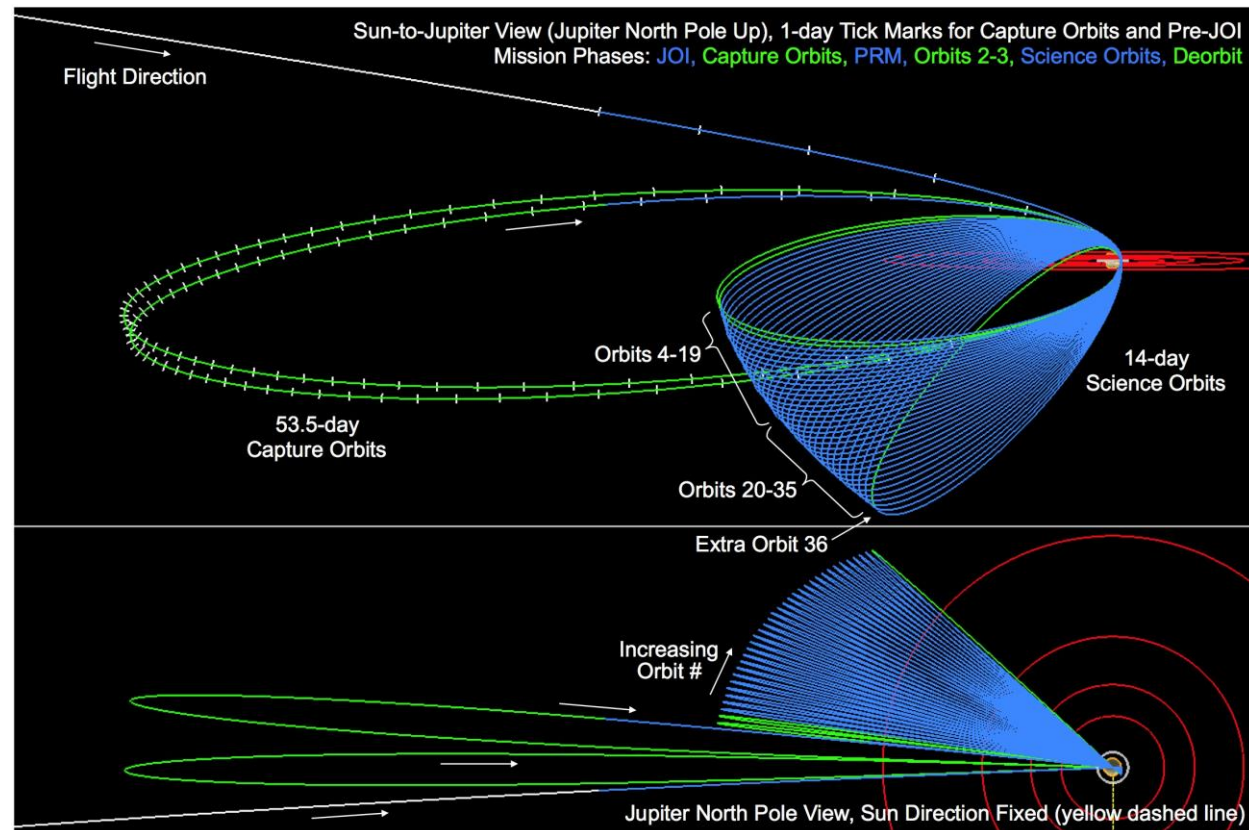
- **Launch: 05-Aug-2011**
- **DSM-1 & 2: 30-Aug & 14-Sep-2012**
 - Deep space maneuver done in two segments.
 - Limited Earth impact probability via a biased aimpoint, off EGA target
- **Trajectory Correction Maneuvers:**
 - **TCM-6,7,8 to aimpoint for EGA**
 - TCM-6 at EGA – 63 days
 - TCM-7 at EGA – 30 days
 - TCM-8 at EGA – 10 days
 - **CAM (collision avoidance maneuver), to avoid a tracked Earth-orbiting object**
 - CAM at EGA – 12 hours
 - **TCM-8 and CAM cancelled**
 - **Only TCM-6 and 7 were executed.**
- **EGA: 09-Oct-2013**
 - **Provided delta-V to get Juno to Jupiter**



Maneuvers: EGA to JOI-CLN

- **EGA: 09-Oct-2013**
- **TCM-9: 13-Nov-2013**
 - EGA clean-up maneuver
- ~~TCM-10: Cancelled~~
- **TCM-11: 03-Feb-2016**
 - Targeting to JOI
- ~~TCM-12: Cancelled~~
- ~~TCM-12a: Cancelled~~
- ~~TCM-13: Cancelled~~
- **JOI: 05-Jul-2016**
 - Capture orbit of 53.5-day period
- **JOI-CLN: 13-Jul-2016**
 - Clean-up errors in flyby & JOI and target PJ01

Mission design at the time of JOI

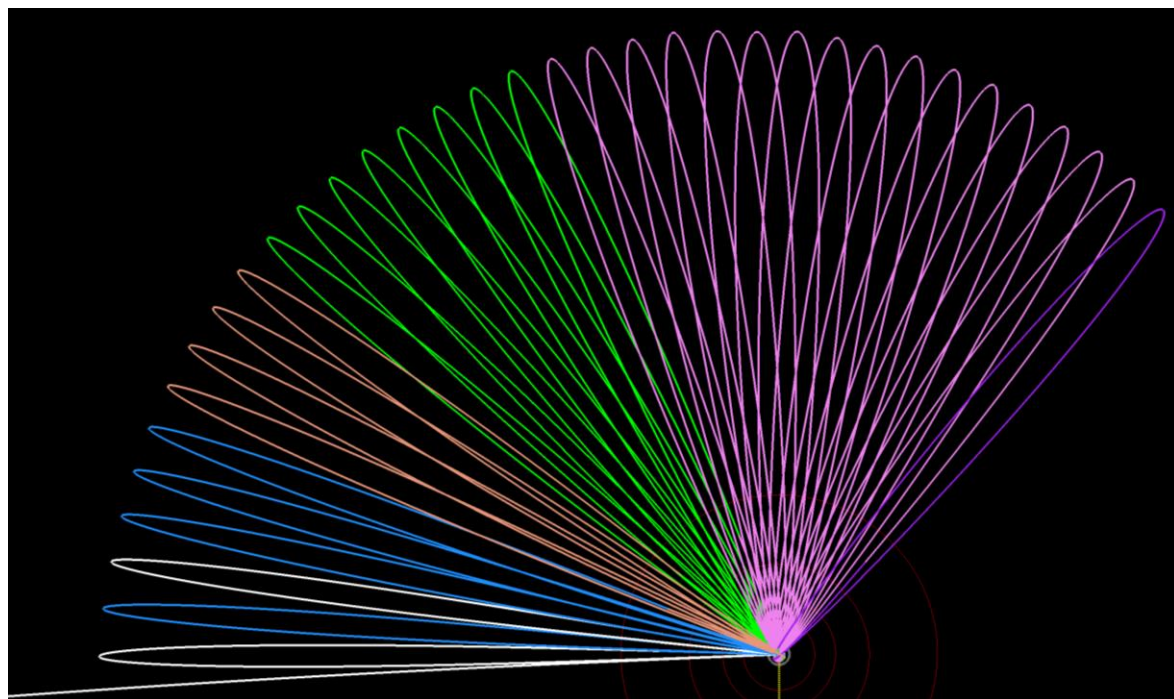


Perijoves & Orbit Trim Maneuvers: PJ01 to PJ06



- PJ01: 27-Aug-2016
 - OTM01(PJ+18days)
- ~~Period Reduction Maneuver (PRM)~~
 - Sticky check-valve
- PJ02: 19-Oct-2016
 - s/c safe mode
 - OTM02(PJ+6days)
- PJ03: 11-Dec-2016
 - OTM03(PJ+7.5hrs)
- PJ04: 02-Feb-2017
 - OTM04(PJ+7.5hrs)
 - APO04(PJ+20days)
- PJ05: 27-Mar-2017
 - OTM05(PJ+7.5hrs)
- PJ06: 19-May-2017
- All STMs were cancelled!

Mission design
post-PRM cancellation

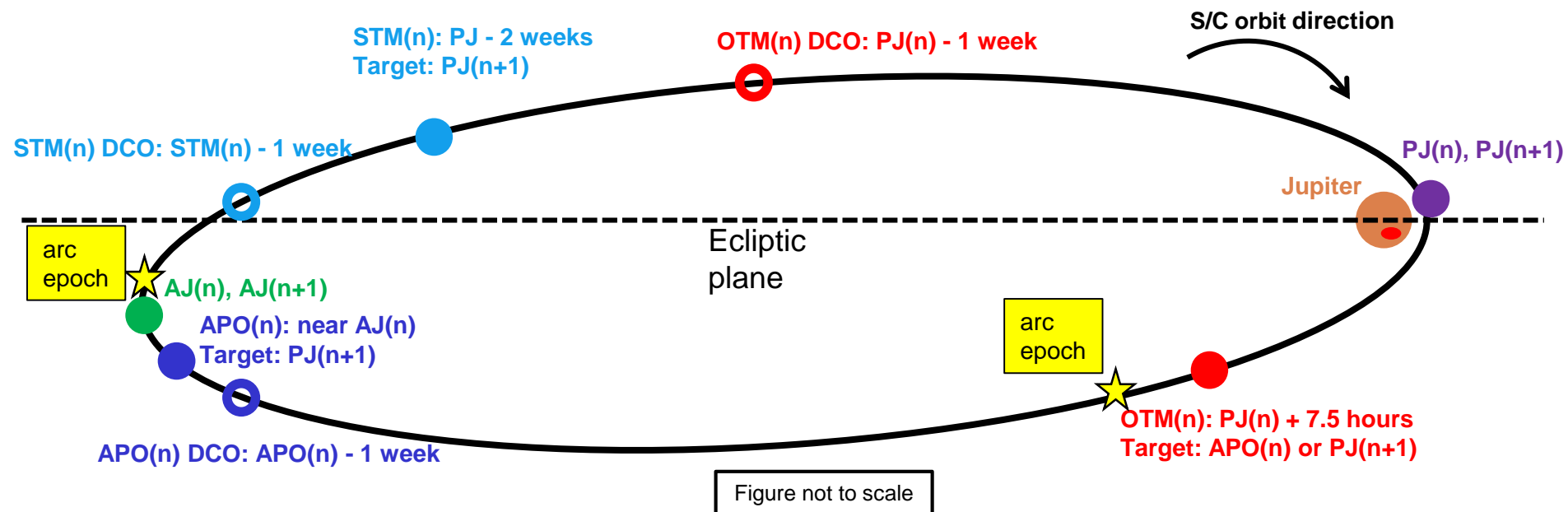


Nav Events During 53-day orbit

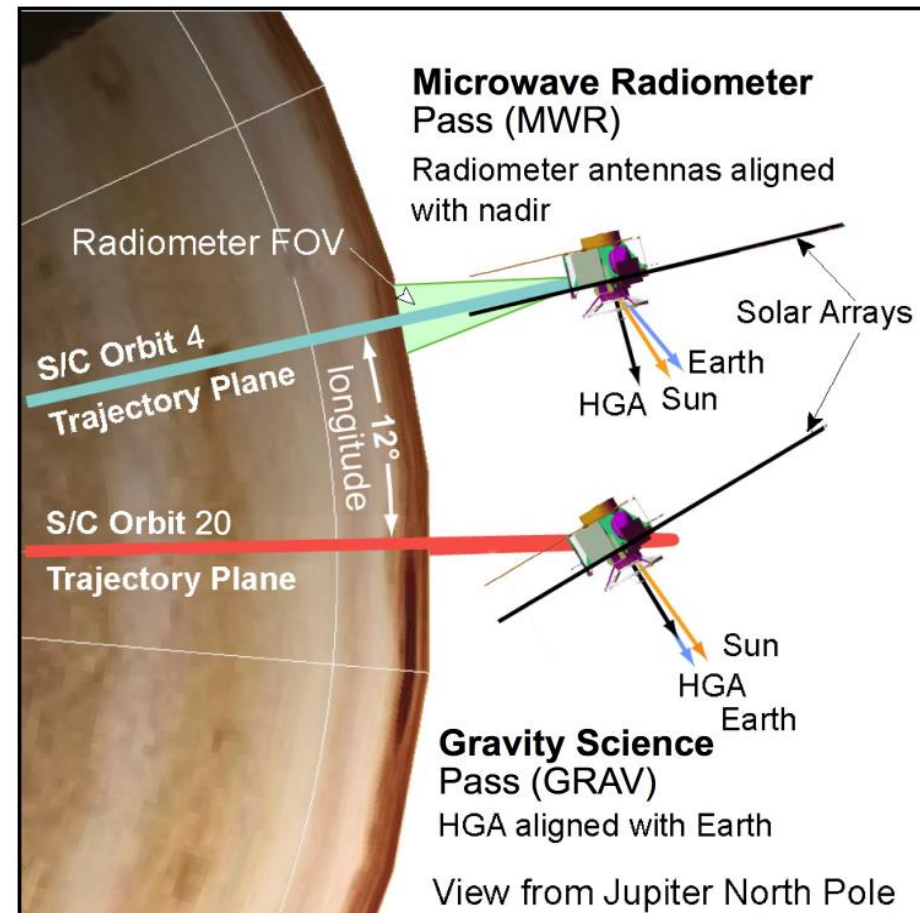
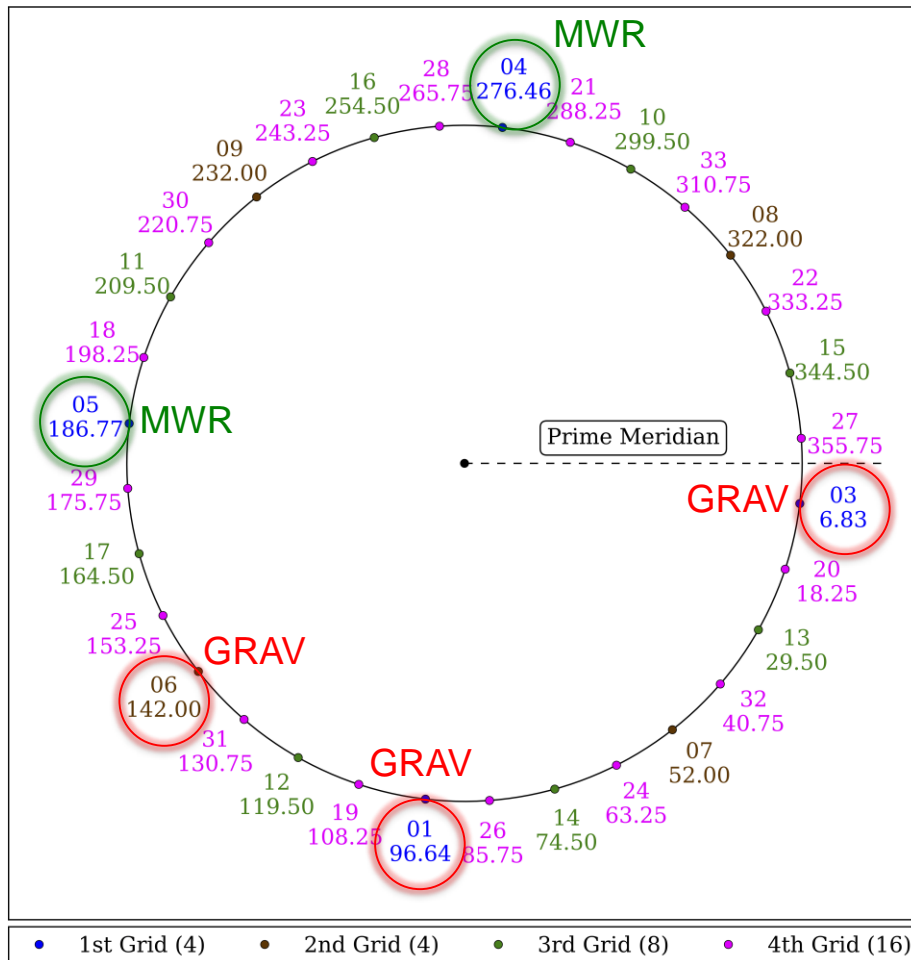
OTM: deterministic **O**rbit **T**rim **M**aneuver

APO: deterministic OTM that occurs near **APO**jove

STM: **S**tatistical [**O**rbit] **T**rim **M**aneuver



Equator-Crossing Longitude Grid and Two Types of Attitudes at Perijove



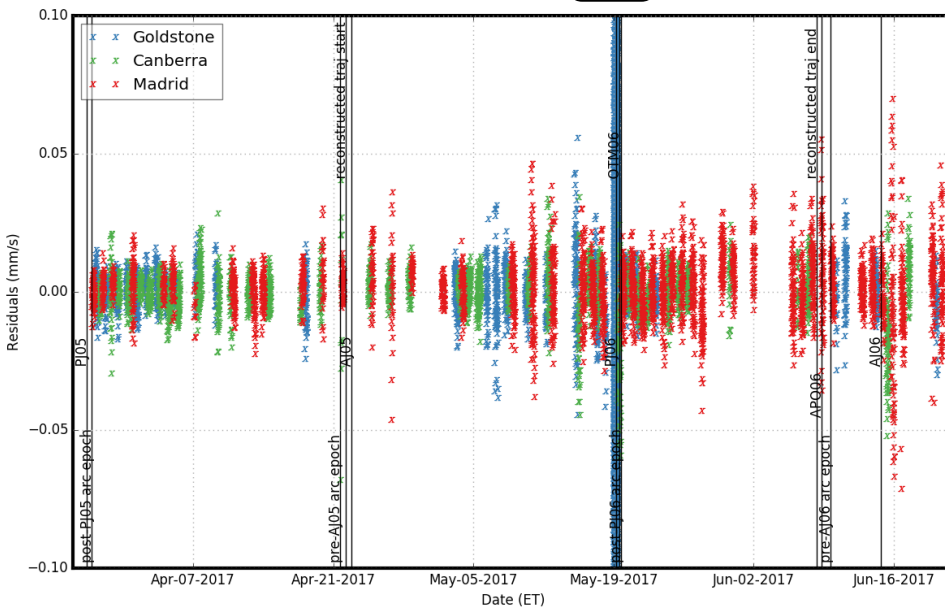
Orbit Determination Setup and Requirements



- **Filter setup varies, but the baseline case estimates:**
 - **State: position & velocity**
 - **Solar Radiation Pressure**
 - **Small forces**
 - **Transponder bias (constant range offset)**
 - **Maneuvers**
- **Side cases include estimating Jupiter barycenter ephemeris, DE436**
 - **transponder bias not estimated**
- **Baseline Ops OD usually ignores the PJ data**
 - **fitting PJ data during reconstruction**
 - **stochastic accelerations occasionally used for reconstruction purpose**
- **Doppler data usually compressed to 300-s**
 - **may chose to use uncompressed 5-s data when reconstructing PJ**
- **Trajectory Prediction Requirement**
 - **equator-crossing longitude target within 1-deg, 3σ ($1\sigma = 0.333$ deg)**
- **Trajectory Reconstruction Requirement**
 - **within 10-km during PJ \pm 3hours**

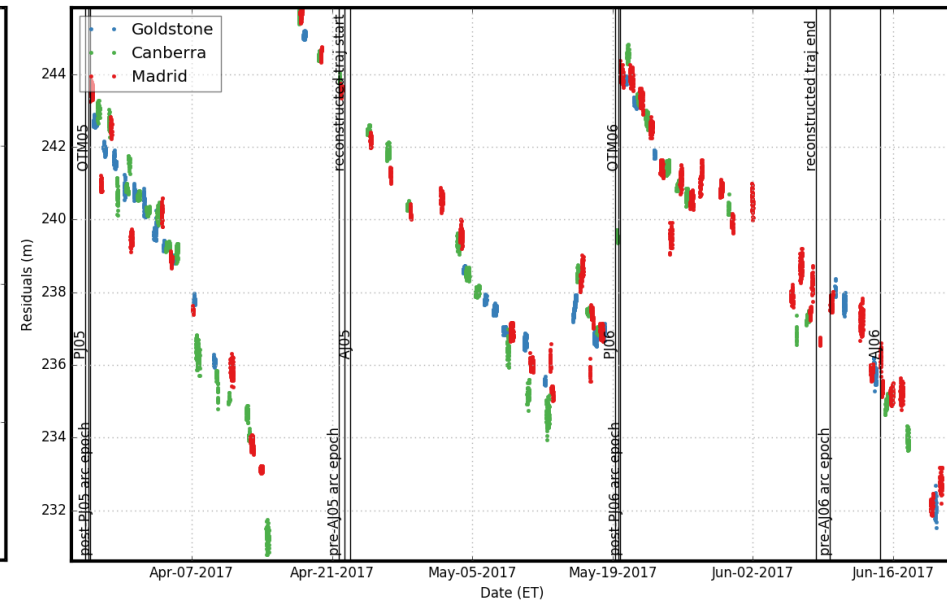
Tracking Data and Fit for Typical Arc

PJ06



2-way Doppler

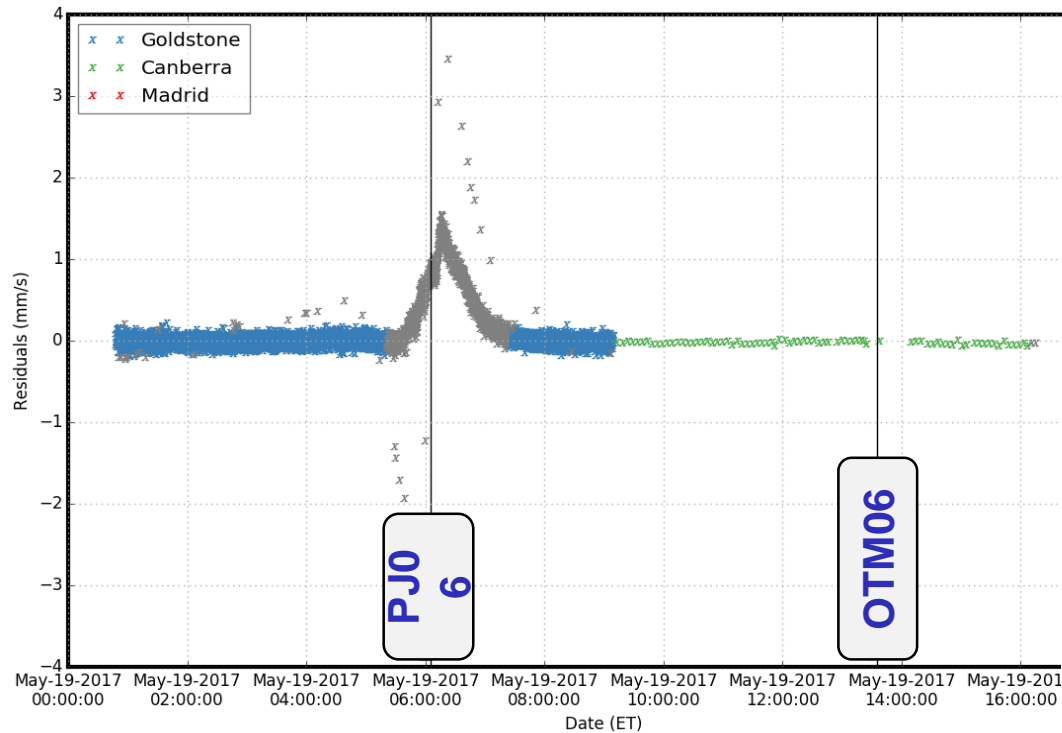
PJ06



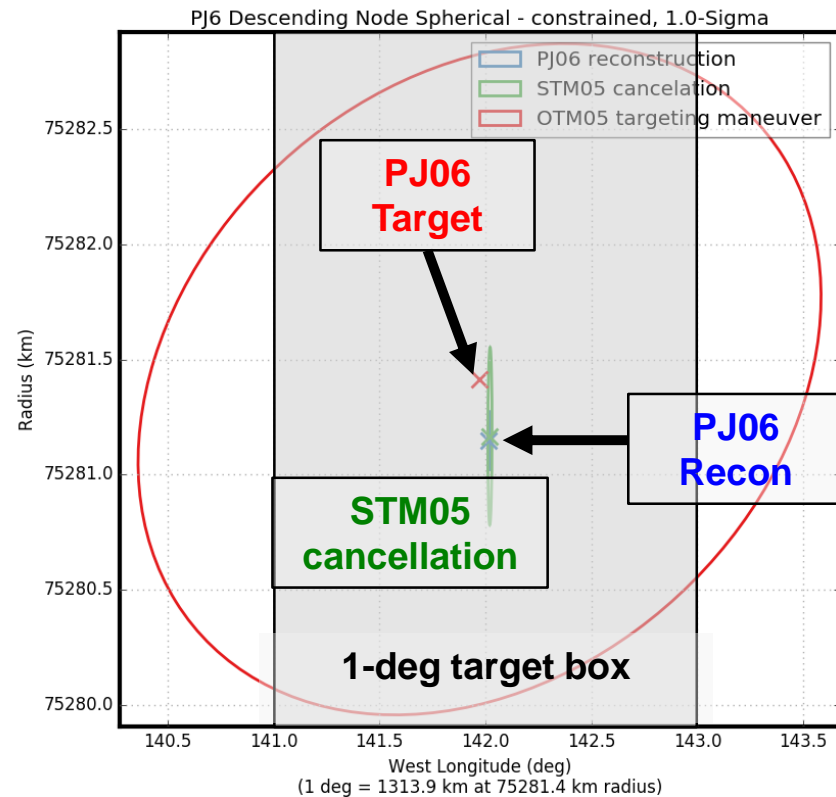
Range



PJ06 Final Results



Doppler residual points in grey
are deleted points (lo plasma torus)
and not a part of the fit

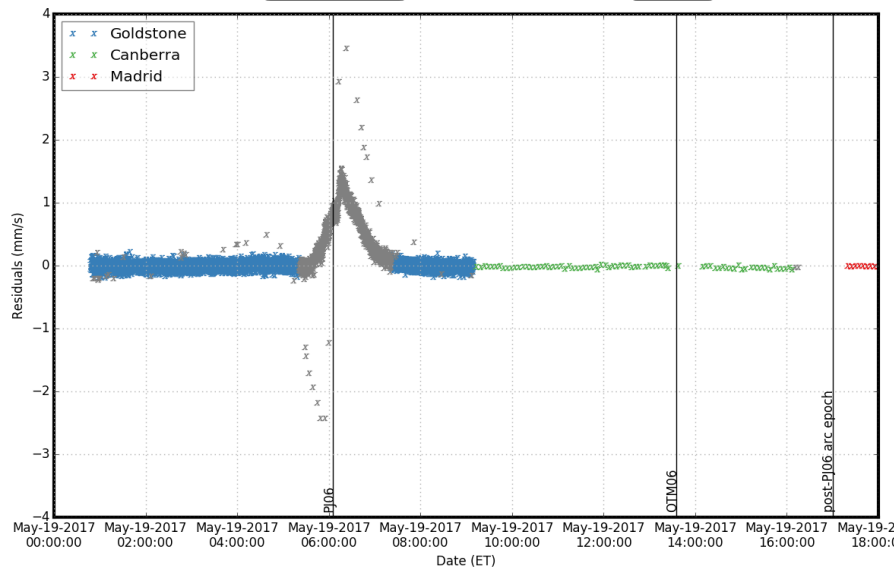


Comparison of PJ05 & PJ06 Final Results



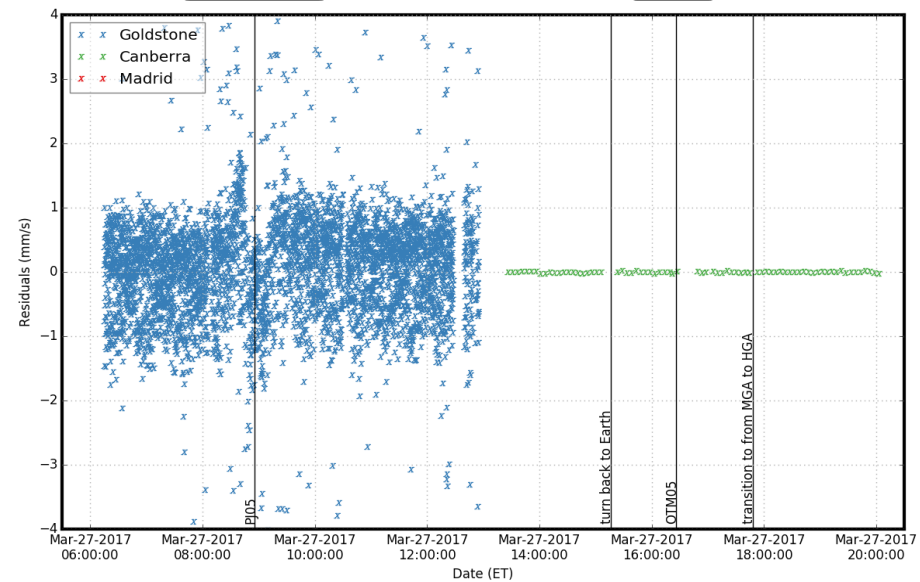
PJ06
on HGA

OTM06



PJ05
on MGA

OTM05



Doppler residual points in grey
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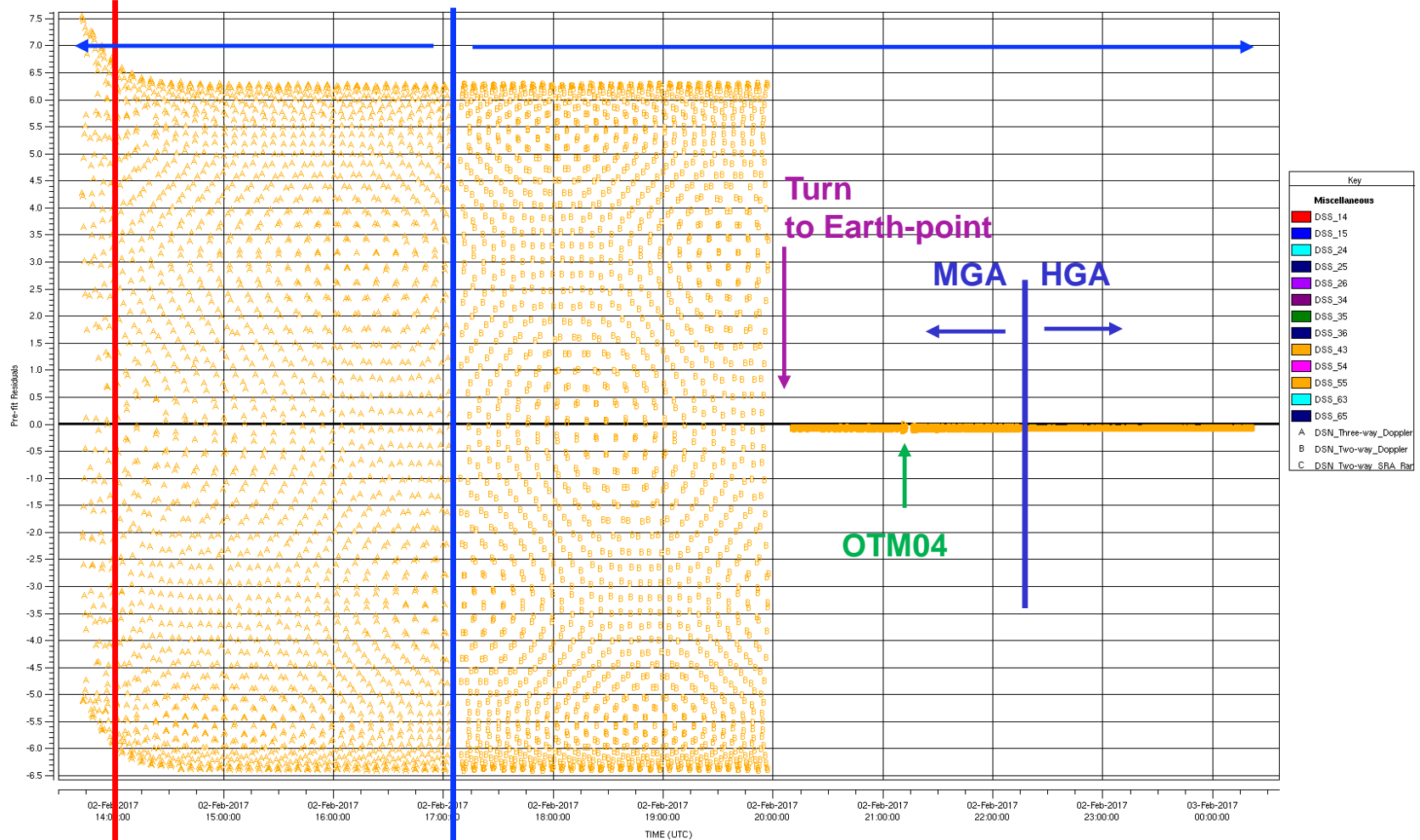


Passthru of Raw PJ4 Data

Three-way: DSS 25 – DSS 43

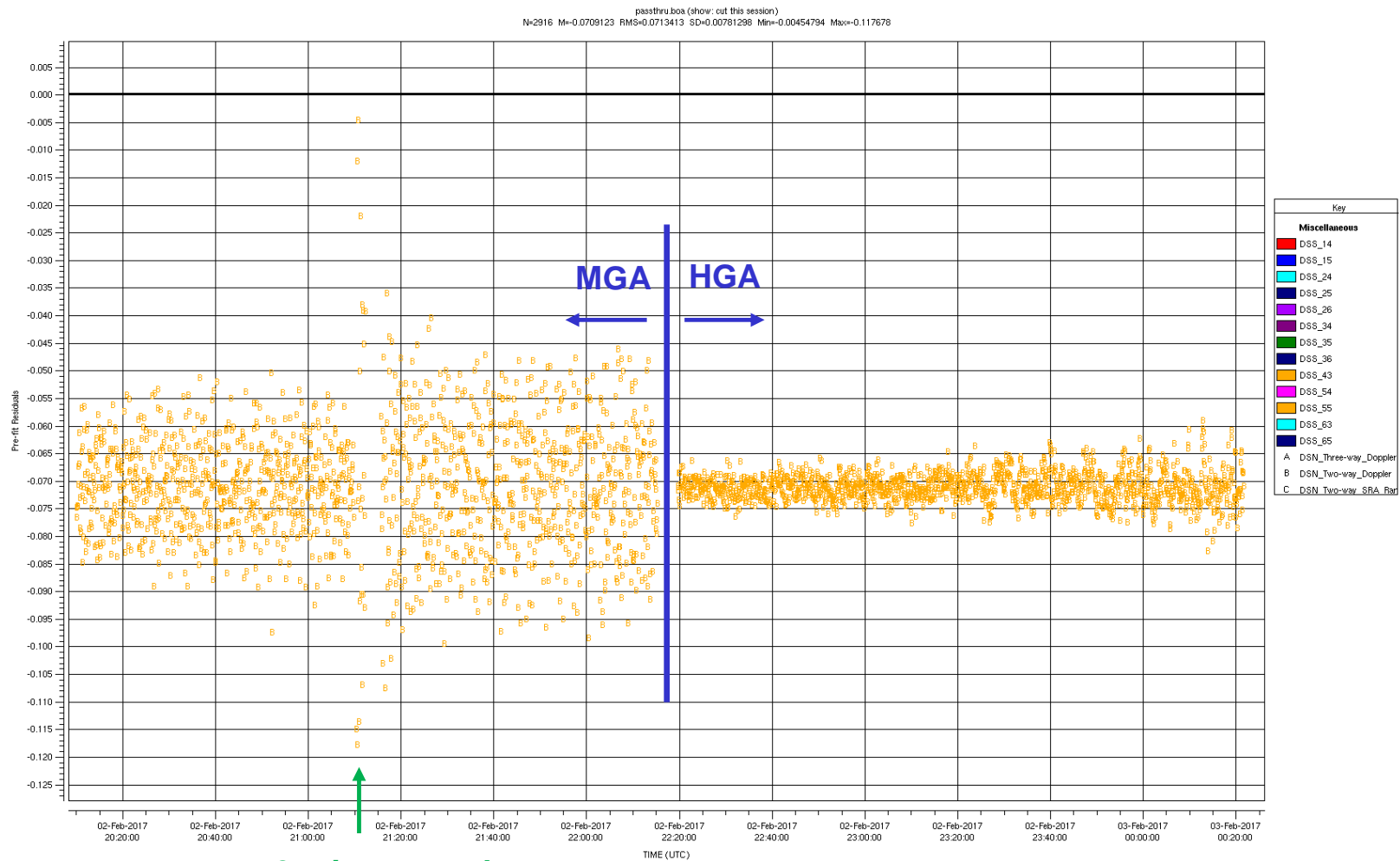
Two-way: DSS 43

κ7.54383



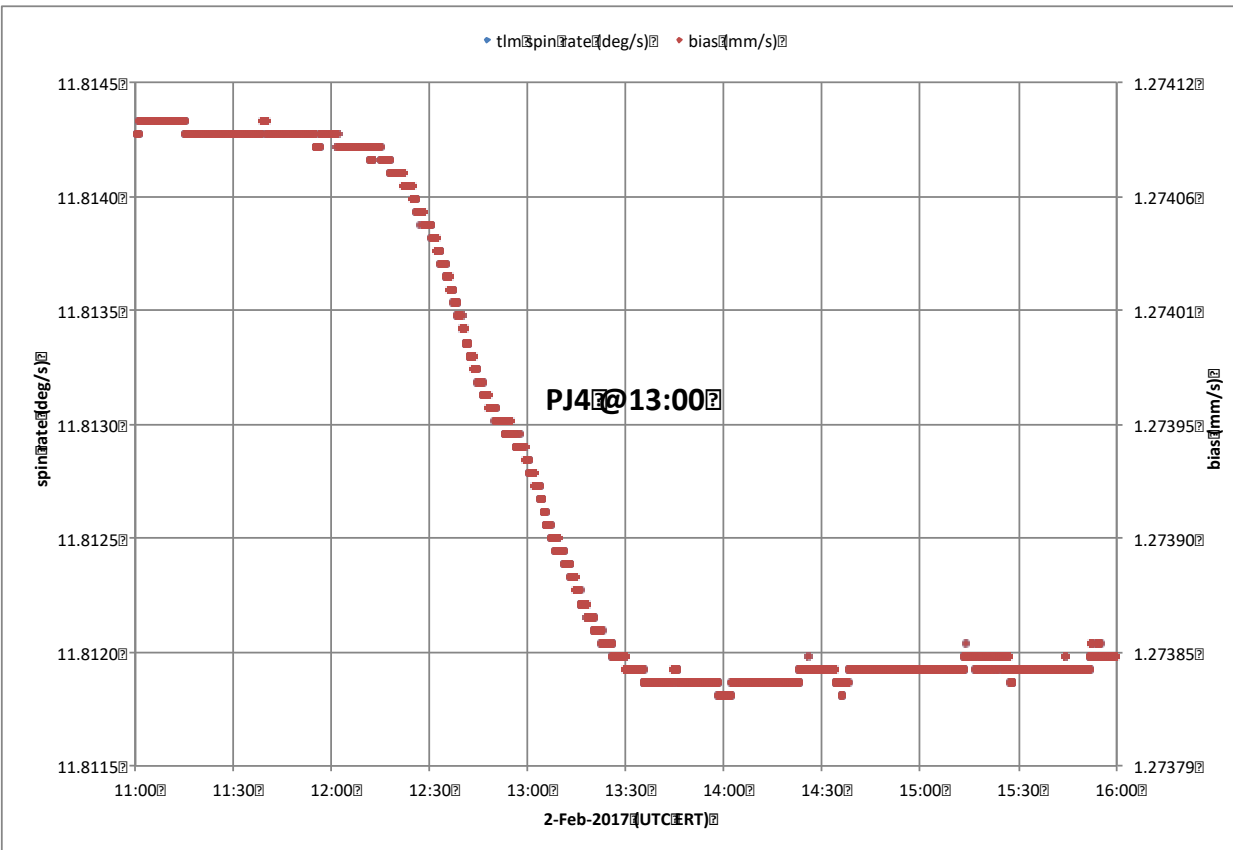
PJ04

Passthru of Raw PJ4 Data zoomed in on Earth-pointed data around OTM04



2-minute gap between
axial and lateral
during OTM04

Spin Rate Change During PJ04

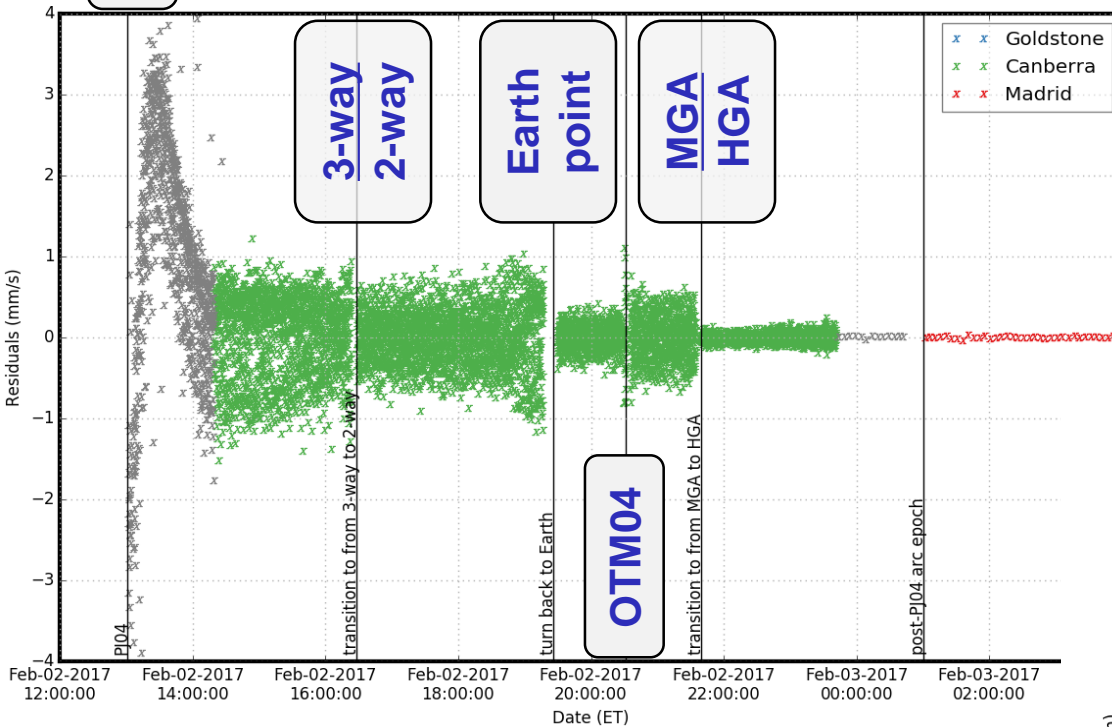


- Spin rate change due to magnetic field torques
- Normally ignore when using HGA during PJ
 - spin-induced bias on Doppler < 0.0005 mm/s
- Can not ignore when using MGA during PJ
 - despin tool assumes near constant spin rate
 - spin rate change too great for tool to determine valid spin state for entire PJ
 - broke up PJ pass into shorter segments
- Alternate method developed to use high-rate spacecraft telemetry for future PJs using MGA

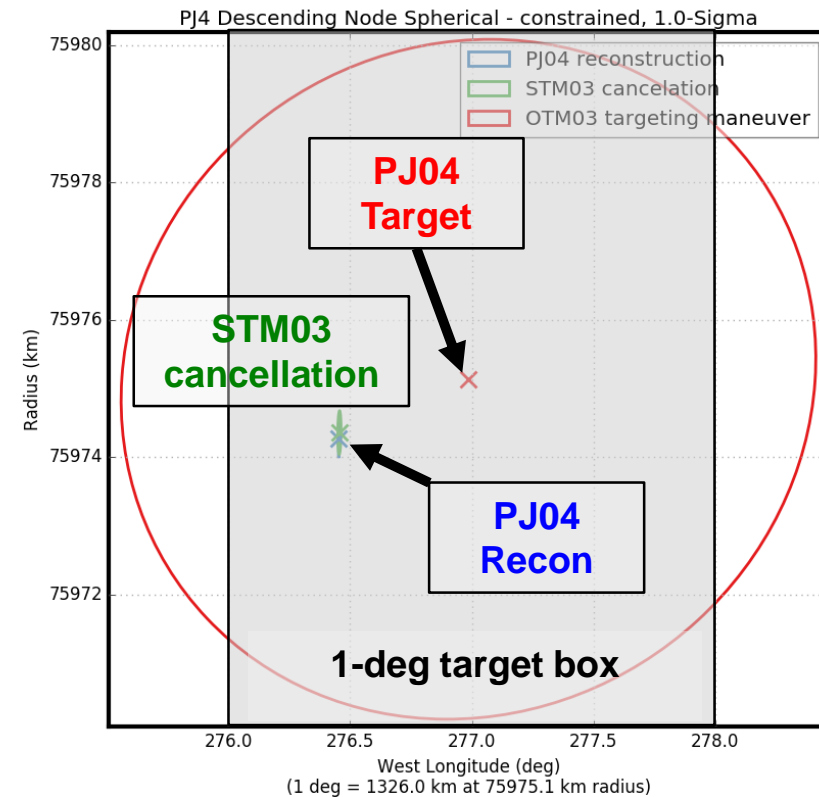


PJ04
4

PJ04 Final Results



Doppler residual points in grey are deleted points (lo plasma torus) and not a part of the fit

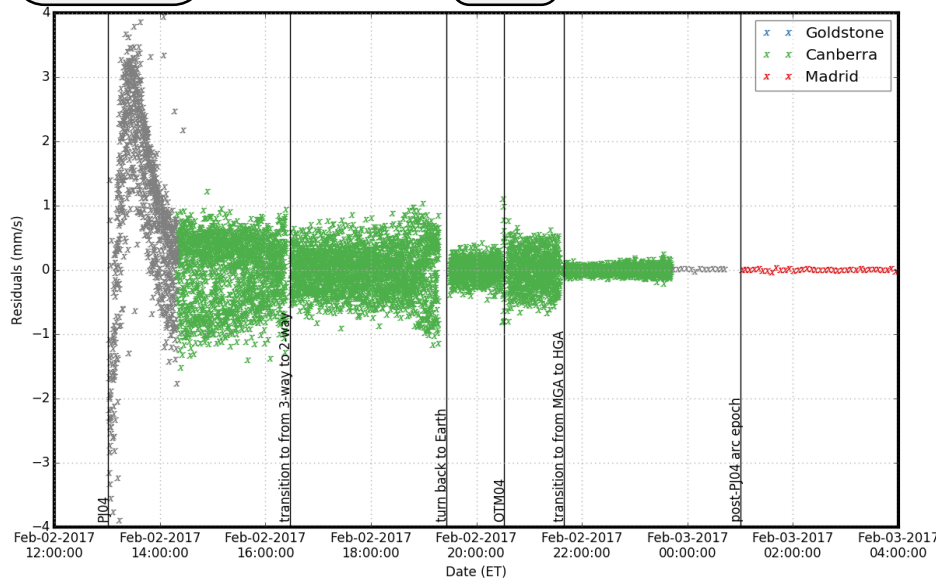


Comparison of PJ04 & PJ05 Final Results



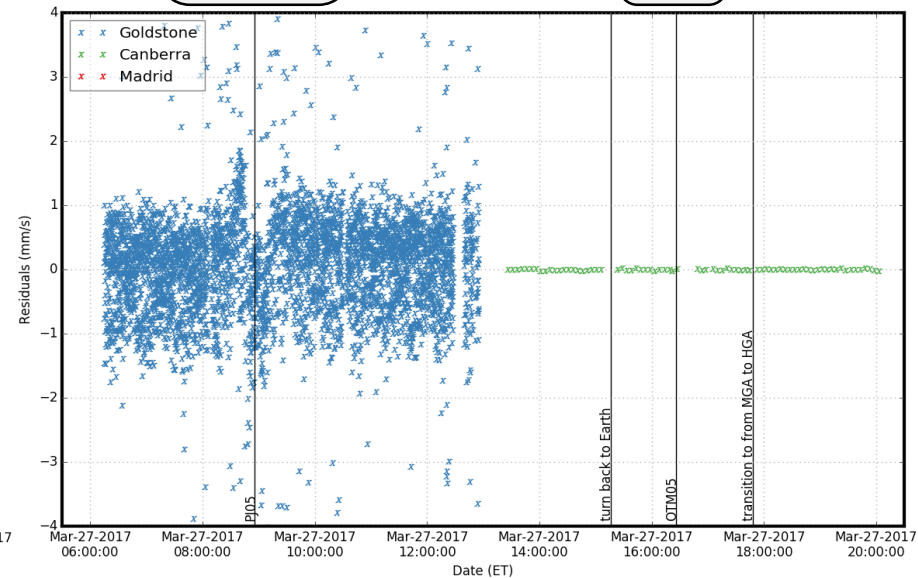
PJ04
old way

OTM04



PJ05
new way

OTM05



Doppler residual points in grey
are deleted points (lo plasma torus)
and not a part of the fit



Summary

- **Juno successfully completed its first year in orbit**
- **Cancellation of PRM led to changes in OD process**
 - additional product deliveries and new data arc strategy
- **All reconstructed trajectories were within 0.5° of equator-crossing longitude target**
 - all statistical maneuvers were cancelled
- **In regular communication with Solar System Dynamics group and Radio Science team**
 - collaborative relationship allowed improvement in reconstructions with better Jupiter barycenter ephemeris, gravity field, and pole orientation
- **Developed alternate method to remove spin signature from Doppler data during PJs using MGA**

Future Outlook



Cloud tops
at 7,800 mi
(12,500 km)

